

Computational Approach to Micromechanical Contacts

Lecture 1.

Introduction to the Finite Element Method

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Elements of tensor algebra

Tensor notations

- Scalars $\in \mathbb{R}$:
 a, α, C

Component notations

- Scalars $\in \mathbb{R}$:
 a, α, C

Tensor notations

- Scalars $\in \mathbb{R}$:
 a, α, C
- Vectors $\in \mathbb{V}_{\text{dim}}$:
 $\underline{a}, \underline{\tau}$

Component notations

- Scalars $\in \mathbb{R}$:
 a, α, C
- Vectors* $\in \mathbb{R}^{\text{dim}}$:
 a_i, τ_j
with $\underline{a} = a_i \underline{e}^i$ and $a_i = \underline{e}^i \cdot \underline{a}$

*Component notations require introducing a basis $\underline{e}^i, i = 1 \dots \text{dim}$ and a dual basis \underline{e}_j such that $\underline{e}_j \cdot \underline{e}^i = \delta_j^i$, where $\delta_j^i = 0$ if $i \neq j$ and $\delta_j^i = 1$ if $i = j$.

Tensor notations

- Scalars $\in \mathbb{R}$:
 a, α, C
- Vectors $\in \mathbb{V}_{\text{dim}}$:
 $\underline{a}, \underline{\tau}$
- Second-order tensors $\in \mathbb{T}_{\text{dim}}^2$:
 $\underline{\underline{A}}, \underline{\underline{\sigma}}$

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with $\underline{a} = a_i \underline{e}^i$ and $a_i = \underline{e}^i \cdot \underline{a}$
- Second-order tensors $\in \mathbb{R}^{\text{dim}} \times \mathbb{R}^{\text{dim}}$:
 A_{ij}, σ_{kl}
with $\underline{\underline{A}} = A_{ij} \underline{e}^i \otimes \underline{e}^j$ and $A_{ij} = \underline{e}_i \cdot \underline{\underline{A}} \cdot \underline{e}_j$

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- Second-order tensors $\in \mathbb{T}_{\text{dim}}^2$:
 $\underline{\underline{A}}, \underline{\underline{\sigma}}$
- Forth-order tensors $\in \mathbb{T}_{\text{dim}}^4$:
 $\underline{\underline{\underline{C}}}$

Component notations

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- Forth-order tensors
 $\in \mathbb{R}^{\text{dim}} \times \dots \times \mathbb{R}^{\text{dim}}$:
 C_{ijkl}
with $\underline{\underline{\underline{C}}} = C_{ijkl} \underline{e}^i \otimes \underline{e}^j \otimes \underline{e}^k \otimes \underline{e}^l$ and
 $C_{ijkl} = \underline{e}_l \cdot (\underline{e}_k \cdot (\underline{e}_j \cdot (\underline{e}_i \cdot \underline{\underline{\underline{C}}}))$

*Component notations require introducing a basis $\underline{e}^i, i = 1 \dots \text{dim}$ and a dual basis \underline{e}_j such that $\underline{e}_j \cdot \underline{e}^i = \delta_j^i$, where $\delta_j^i = 0$ if $i \neq j$ and $\delta_j^i = 1$ if $i = j$.

Tensor notations

- Transposition

$$\underline{\underline{C}} = \underline{\underline{D}}^\top, (\underline{\underline{A}} \cdot \underline{\underline{B}})^\top = \underline{\underline{B}}^\top \cdot \underline{\underline{A}}^\top$$

- Symmetric tensor

$$\underline{\underline{A}}^\top = \underline{\underline{A}}$$

- Antisymmetric* tensor

$$\underline{\underline{B}}^\top = -\underline{\underline{B}}$$

- Tensor decomposition

$$\underline{\underline{C}} = \underline{\underline{C}}^S + \underline{\underline{C}}^A \quad \text{with} \\ \underline{\underline{C}}^S = \frac{1}{2} (\underline{\underline{C}} + \underline{\underline{C}}^\top), \underline{\underline{C}}^A = \frac{1}{2} (\underline{\underline{C}} - \underline{\underline{C}}^\top)$$

Component notations

- Transposition

$$C_{ij} = D_{ji}$$

- Symmetric tensor

$$A_{ij} = A_{ji}$$

- Antisymmetric tensor

$$B_{ij} = -B_{ji}$$

- Tensor decomposition

$$C_{ij} = C_{ij}^S + C_{ij}^A \quad \text{with} \\ C_{ij}^S = \frac{1}{2} (C_{ij} + C_{ji}), C_{ij}^A = \frac{1}{2} (C_{ij} - C_{ji})$$

Examples

- Identity tensor (symmetric) $\underline{\underline{I}} = \delta^{ij} \underline{\underline{e}}_i \otimes \underline{\underline{e}}_j = \underline{\underline{e}}_i \otimes \underline{\underline{e}}_i$

- Rotation tensor (asymmetric = symmetric($\neq 0$) + antisymmetric($\neq 0$)):

$$\underline{\underline{Q}} \sim \begin{bmatrix} \cos(\phi) & \sin(\phi) & 0 \\ -\sin(\phi) & \cos(\phi) & 0 \\ 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} \cos(\phi) & 0 & 0 \\ 0 & \cos(\phi) & 0 \\ 0 & 0 & 1 \end{bmatrix} + \begin{bmatrix} 0 & \sin(\phi) & 0 \\ -\sin(\phi) & 0 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

*Antisymmetric \equiv skew-symmetric

Tensor algebra: products

Tensor notations

- Multiplication by a scalar

$$\underline{\underline{\alpha}} \underline{\underline{A}} = \underline{\underline{A}} \underline{\underline{\alpha}}$$

- Scalar product*

$$\underline{a} \cdot \underline{b} = c$$

$$\underline{a} \cdot \underline{\underline{A}} = \underline{b}$$

$$\underline{\underline{A}} \cdot \underline{\underline{B}} = \underline{\underline{C}}$$

- Tensor contraction

$$\underline{\underline{A}} : \underline{\underline{B}} = c$$

$$\underline{\underline{A}} \cdot \cdot \underline{\underline{B}} = d$$

- Remark:

$$\underline{\underline{A}} : \underline{\underline{B}} = \underline{\underline{A}}^S : \underline{\underline{B}}^S + \underline{\underline{A}}^A : \underline{\underline{B}}^A \text{ and } \underline{\underline{A}}^S : \underline{\underline{B}}^A = \underline{\underline{A}}^A : \underline{\underline{B}}^S = 0$$

$$\underline{\underline{A}} \cdot \cdot \underline{\underline{B}} = \underline{\underline{A}}^S \cdot \cdot \underline{\underline{B}}^S + \underline{\underline{A}}^A \cdot \cdot \underline{\underline{B}}^A \text{ and } \underline{\underline{A}}^S \cdot \cdot \underline{\underline{B}}^A = \underline{\underline{A}}^A \cdot \cdot \underline{\underline{B}}^S = 0$$

Component notations

- Multiplication by a scalar

$$\alpha A_{ij} = A_{ij} \alpha$$

- Scalar (dot) product**

$$a_i b^i = c$$

$$a_i A^{ij} = b^j$$

$$A_{ij} B^{jk} = C_i^k$$

- Tensor contraction

$$A_{ij} B_{ij} = c$$

$$A_{ij} B_{ji} = d$$

*Scalar product \equiv dot product \equiv inner product.

**We assume Einstein summation by repeating index, i.e. $a_i A^{ij} = \sum_{i=1}^{\dim} a_i A^{ij}$.

Tensor algebra: products II & invariants

Tensor notations

- Vector product*

$$\underline{a} \times \underline{b} = \underline{c}$$

such that $\underline{c} \cdot \underline{a} = 0, \underline{c} \cdot \underline{b} = 0$

- Tensor product**

$$\underline{a} \otimes \underline{b} = \underline{\underline{C}}$$

$$\underline{\underline{A}} \otimes \underline{\underline{B}} = \underline{\underline{\underline{C}}}$$

- Invariants:

$$I_1(\underline{\underline{A}}) = \text{tr}(\underline{\underline{A}}) = \underline{\underline{I}} : \underline{\underline{A}}$$

$$I_2(\underline{\underline{A}}) = \frac{1}{2} [\text{tr}(\underline{\underline{A}})^2 - \text{tr}(\underline{\underline{A}}^2)]$$

$$I_3(\underline{\underline{A}}) = \det(\underline{\underline{A}})$$

Component notations

- Vector product*

$$c^i = \epsilon_{ijk} a^j b^k$$

with ϵ_{ijk} Levi-Civita symbol

$$\epsilon_{ijk} = \begin{cases} 1, & \text{if } (i, j, k) = (1, 2, 3) \text{ or } (2, 3, 1) \text{ or } (3, 1, 2) \\ -1, & \text{if } (i, j, k) = (2, 1, 3) \text{ or } (1, 3, 2) \text{ or } (3, 2, 1) \\ 0, & \text{otherwise.} \end{cases}$$

- Tensor product

$$a_i b_j = C_{ij}$$

$$A_{ij} B_{kl} = C_{ijkl}$$

- Invariants:

$$I_1(\underline{\underline{A}}) = A_{ii} = A_{11} + A_{22} + A_{33}$$

$$I_2(\underline{\underline{A}}) = \dots$$

$$I_3(\underline{\underline{A}}) = \dots$$

*Defined only for $\dim = 3$, also called cross product.

**Also called outer product.

Tensor algebra: deviatoric & spherical parts

Tensor notations

- Spherical part of tensor $\underline{\underline{A}}$

$$\text{Sp}(\underline{\underline{A}}) = \frac{1}{3} \text{tr}(\underline{\underline{A}}) \underline{\underline{I}}$$

- Deviatoric part of tensor $\underline{\underline{A}}$

$$\text{Dv}(\underline{\underline{A}}) = \underline{\underline{A}} - \frac{1}{3} \text{tr}(\underline{\underline{A}}) \underline{\underline{I}}$$

- Tensor decomposition

$$\underline{\underline{A}} = \text{Sp}(\underline{\underline{A}}) + \text{Dv}(\underline{\underline{A}})$$

- Remark: for an antisymmetric tensor $\underline{\underline{B}}^A$

$$\text{Sp}(\underline{\underline{B}}^A) = 0 \quad \Rightarrow \quad \underline{\underline{B}}^A = \text{Dv}(\underline{\underline{B}}^A)$$

Component notations

- Spherical part of tensor $\underline{\underline{A}}$

$$\text{Sp}(\underline{\underline{A}}) = \frac{1}{3} (A_{kk}) \delta_{ij}$$

- Deviatoric part of tensor $\underline{\underline{A}}$

$$\text{Dv}(\underline{\underline{A}}) = A_{ij} - \frac{1}{3} (A_{kk}) \delta_{ij}$$

Tensor algebra: principal values

- Principal values of a linear operator $\underline{\underline{A}}$:

$$\underline{\underline{A}} \cdot \underline{u} = \lambda \underline{u} \quad \Leftrightarrow \quad (\underline{\underline{A}} - \lambda \underline{\underline{I}}) \cdot \underline{u} = 0$$

If $\underline{\underline{A}} = \underline{\underline{A}}^S$ for $\dim = 3$ then exist three real λ_i and corresponding \underline{u}_i called eigen values and eigen vectors of operator $\underline{\underline{A}}$, respectively. Moreover, for $i \neq j$, $\underline{u}_i \cdot \underline{u}_j = 0$.

- To find λ_i we solve

$$I_3(\underline{\underline{A}}) - I_2(\underline{\underline{A}})\lambda + I_1(\underline{\underline{A}})\lambda^2 - \lambda^3 = 0$$

- Then tensor can be rewritten in its eigen basis:

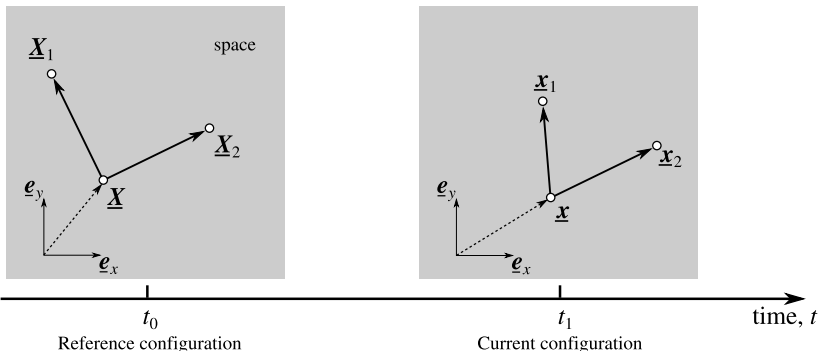
$$\underline{\underline{A}} = \lambda_1 \underline{u}_1 \otimes \underline{u}_1 + \lambda_2 \underline{u}_2 \otimes \underline{u}_2 + \lambda_3 \underline{u}_3 \otimes \underline{u}_3$$

and $\text{tr}(\underline{\underline{A}}) = \lambda_i |\underline{u}_i|^2$.

Continuum Mechanics: Recall

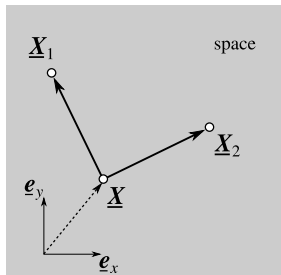
Deformable medium

- Consider change in positions of points with time t
- Consider two states: $t = t_0$ (reference) and $t = t_1$ (current configurations)
- Point \underline{X} from the reference configuration is labeled \underline{x} in the current configuration
- Displacement vector between t_0 and t_1 is $\underline{u} = \underline{x} - \underline{X}$

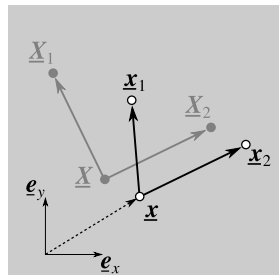


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t_0
Reference configuration

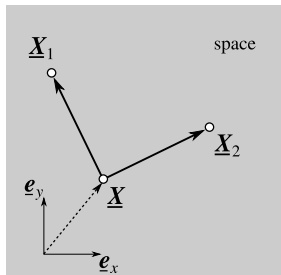


t_1
Current configuration

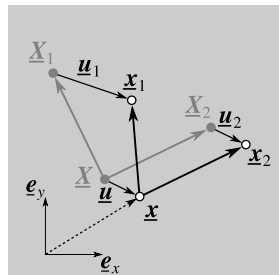
time, t

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t_0
Reference configuration



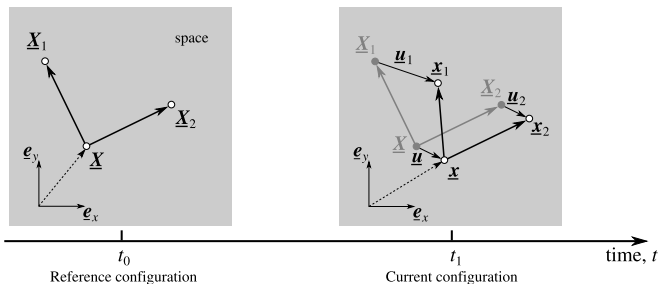
t_1
Current configuration

time, t

Deformation tensor

- Transformation gradient $\underline{\underline{F}} = \frac{\partial \underline{x}}{\partial \underline{X}} = \frac{\partial(\underline{X} + \underline{u})}{\partial \underline{X}} = \underline{\underline{I}} + \frac{\partial \underline{u}}{\partial \underline{X}} = \underline{\underline{I}} + \underline{\underline{H}}$
- Cauchy-Green right tensor $\underline{\underline{C}} = \underline{\underline{F}}^T \cdot \underline{\underline{F}}$
- Green-Lagrange deformation tensor $\underline{\underline{E}} = \frac{1}{2} (\underline{\underline{C}} - \underline{\underline{I}}) = \underline{\underline{H}}^S + \frac{1}{2} \underline{\underline{H}}^T \cdot \underline{\underline{H}}$
- For $H_{ij} \ll 1$, $\underline{\underline{E}} \approx \underline{\underline{H}}^S$ and we obtain a tensor of small deformations

$$\underline{\underline{\varepsilon}} = \underline{\underline{H}}^S = \frac{1}{2} \left[\frac{\partial \underline{u}}{\partial \underline{X}} + \left(\frac{\partial \underline{u}}{\partial \underline{X}} \right)^T \right] = \frac{1}{2} (\nabla \underline{u} + (\nabla \underline{u})^T)$$



Stress tensor and Hooke's law

- Hooke's law in uniaxial test:

$$\sigma_{xx} = E \varepsilon_{xx}$$

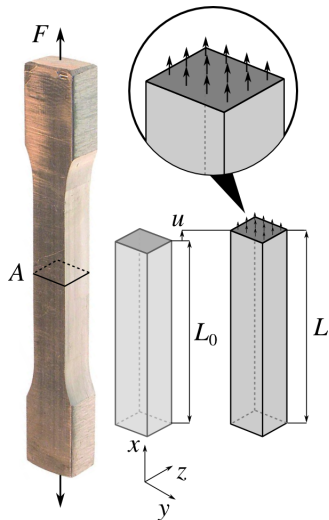
$$F = ku \Leftrightarrow \sigma_{xx} A = \frac{EA}{L_0} u = EA \frac{L - L_0}{L_0}$$

- In general case stress and strain are related through a linear operator (fourth-order elasticity tensor $\underline{\underline{C}}$):

$$\underline{\underline{\sigma}} = \underline{\underline{C}} : \underline{\underline{\varepsilon}}$$

- Inversely the strain can be found through a stiffness tensor $\underline{\underline{S}}$:

$$\underline{\underline{\varepsilon}} = \underline{\underline{S}} : \underline{\underline{\sigma}}$$



Hooke's law for isotropic solids: stress

- In the case of isotropic material the Hooke's law reduces to:

$$\underline{\underline{\sigma}} = \lambda \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{I}} + 2\mu \underline{\underline{\epsilon}},$$

with λ, μ being Lamé coefficients:

$$\lambda = \frac{\nu E}{(1 + \nu)(1 - 2\nu)}, \quad \mu = \frac{E}{2(1 + \nu)}$$

with Young's modulus E and Poisson's ratio ν .

- In the component form it reads:

$$\sigma_{ij} = \lambda(\epsilon_{kk})\delta_{ij} + 2\mu\epsilon_{ij}$$

- In the matrix form:

$$\begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{12} & \sigma_{22} & \sigma_{23} \\ \sigma_{13} & \sigma_{23} & \sigma_{33} \end{bmatrix} = 2\mu \begin{bmatrix} \lambda \text{tr}(\underline{\underline{\epsilon}})/(2\mu) + \epsilon_{11} & \epsilon_{12} & \epsilon_{13} \\ \epsilon_{12} & \lambda \text{tr}(\underline{\underline{\epsilon}})/(2\mu) + \epsilon_{22} & \epsilon_{23} \\ \epsilon_{13} & \epsilon_{23} & \lambda \text{tr}(\underline{\underline{\epsilon}})/(2\mu) + \epsilon_{33} \end{bmatrix}$$

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Hooke's law for isotropic solids: strain

- Strain as a function of stress:

$$\underline{\underline{\varepsilon}} = \frac{1 + \nu}{E} \underline{\underline{\sigma}} - \frac{\nu}{E} \text{tr}(\underline{\underline{\sigma}}) \underline{\underline{I}}.$$

- In the component form it reads:

$$\varepsilon_{ij} = \frac{1 + \nu}{E} \sigma_{ij} - \frac{\nu}{E} \sigma_{kk} \delta_{ij}$$

- In the matrix form:

$$\begin{bmatrix} \varepsilon_{11} & \varepsilon_{12} & \varepsilon_{13} \\ \varepsilon_{12} & \varepsilon_{22} & \varepsilon_{23} \\ \varepsilon_{13} & \varepsilon_{23} & \varepsilon_{33} \end{bmatrix} = \frac{1}{E} \begin{bmatrix} (1 + \nu)\sigma_{11} - \nu \text{tr}(\underline{\underline{\sigma}}) & (1 + \nu)\sigma_{12} & (1 + \nu)\sigma_{13} \\ (1 + \nu)\sigma_{12} & (1 + \nu)\sigma_{22} - \nu \text{tr}(\underline{\underline{\sigma}}) & (1 + \nu)\sigma_{23} \\ (1 + \nu)\sigma_{13} & (1 + \nu)\sigma_{23} & (1 + \nu)\sigma_{33} - \nu \text{tr}(\underline{\underline{\sigma}}) \end{bmatrix}$$

Hooke's law for isotropic solids: strain

- Strain as a function of stress:

$$\underline{\underline{\varepsilon}} = \frac{1 + \nu}{E} \underline{\underline{\sigma}} - \frac{\nu}{E} \text{tr}(\underline{\underline{\sigma}}) \underline{\underline{I}}.$$

- In the component form it reads:

$$\varepsilon_{ij} = \frac{1 + \nu}{E} \sigma_{ij} - \frac{\nu}{E} \sigma_{kk} \delta_{ij}$$

- In the matrix form:

$$\begin{bmatrix} \varepsilon_{11} & \varepsilon_{12} & \varepsilon_{13} \\ \varepsilon_{12} & \varepsilon_{22} & \varepsilon_{23} \\ \varepsilon_{13} & \varepsilon_{23} & \varepsilon_{33} \end{bmatrix} = \frac{1}{E} \begin{bmatrix} (1 + \nu)\sigma_{11} - \nu \text{tr}(\underline{\underline{\sigma}}) & (1 + \nu)\sigma_{12} & (1 + \nu)\sigma_{13} \\ (1 + \nu)\sigma_{12} & (1 + \nu)\sigma_{22} - \nu \text{tr}(\underline{\underline{\sigma}}) & (1 + \nu)\sigma_{23} \\ (1 + \nu)\sigma_{13} & (1 + \nu)\sigma_{23} & (1 + \nu)\sigma_{33} - \nu \text{tr}(\underline{\underline{\sigma}}) \end{bmatrix}$$
$$= \frac{1}{E} \begin{bmatrix} \sigma_{11} - \nu(\sigma_{22} + \sigma_{33}) & (1 + \nu)\sigma_{12} & (1 + \nu)\sigma_{13} \\ (1 + \nu)\sigma_{12} & \sigma_{22} - \nu(\sigma_{11} + \sigma_{33}) & (1 + \nu)\sigma_{23} \\ (1 + \nu)\sigma_{13} & (1 + \nu)\sigma_{23} & \sigma_{33} - \nu(\sigma_{11} + \sigma_{22}) \end{bmatrix}$$

Equilibrium of an infinitesimal element

- Infinitesimal strain tensor is symmetric and satisfies the compatibility conditions*:

$$\nabla \times (\nabla \times \underline{\underline{\varepsilon}}) = 0$$

- Stress tensor $\underline{\underline{\sigma}}$ should ensure equilibrium of infinitesimal element**:

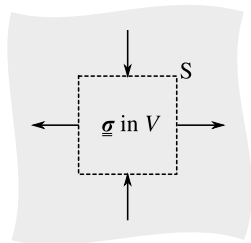
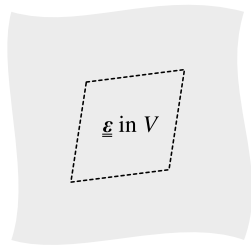
$$\text{Force balance: } \int_S \underline{\underline{n}} \cdot \underline{\underline{\sigma}} dS = 0$$

$$\text{Momentum balance: } \int_S \underline{\underline{r}} \times (\underline{\underline{n}} \cdot \underline{\underline{\sigma}}) dS = 0$$

- Following Gauss-Ostrogradsky theorem:

$$\int_S \underline{\underline{n}} \cdot \underline{\underline{\sigma}} dS = \int_V \nabla \cdot \underline{\underline{\sigma}} dV = 0 \text{ Since volume } V \text{ can be arbitrary chosen, then}$$

$$\boxed{\nabla \cdot \underline{\underline{\sigma}} = 0} \text{ everywhere in } V.$$



*In case of a simply-connected solid.

**In absence of volumetric forces.

Equilibrium of an infinitesimal element II

- Second Newton's law:

$$m\ddot{\underline{u}} = \underline{f} \quad \Rightarrow \quad \rho\ddot{\underline{u}} = \frac{1}{V}\underline{f}$$

- In presence of volumetric forces with density \underline{f}_{-V} , the total force is given by:

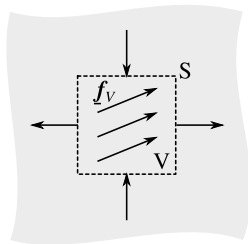
$$\underline{f} = \int_V \underline{f}_{-V} dV + \int_S \underline{n} \cdot \underline{\underline{\sigma}} dS$$

- Then using the second Newton's law and Gauss-Ostrogradsky's theorem:

$$\int_V \left(\nabla \cdot \underline{\underline{\sigma}} + \underline{f}_{-V} \right) dV = \int_V \rho\ddot{\underline{u}} dV$$

- Since it is right for arbitrary V , then in every point of V :

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{f}_{-V} = \rho\ddot{\underline{u}}$$



Equilibrium of an infinitesimal element II

- Equilibrium (3 equations):

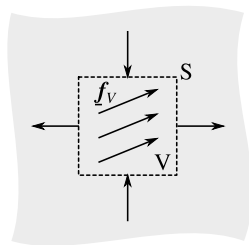
$$\nabla \cdot \underline{\underline{\sigma}} + \underline{f}_{-V} = \rho \underline{\underline{ü}}$$

- In component form:

$$\frac{\partial \sigma_{11}}{\partial x} + \frac{\partial \sigma_{12}}{\partial y} + \frac{\partial \sigma_{13}}{\partial z} + f_{V_x} = \rho \ddot{u}_x$$

$$\frac{\partial \sigma_{12}}{\partial x} + \frac{\partial \sigma_{22}}{\partial y} + \frac{\partial \sigma_{23}}{\partial z} + f_{V_y} = \rho \ddot{u}_y$$

$$\frac{\partial \sigma_{13}}{\partial x} + \frac{\partial \sigma_{23}}{\partial y} + \frac{\partial \sigma_{33}}{\partial z} + f_{V_z} = \rho \ddot{u}_z$$



Deformable solid and boundary conditions

Notations:

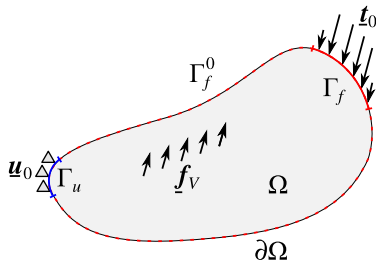
- Consider a solid Ω with boundary $\partial\Omega$
- Boundary is split into Γ_u and Γ_f :
 $\partial\Omega = \Gamma_u \cup \Gamma_f$
- At Γ_u displacements $\underline{u}_0(t, \underline{X})$ are prescribed (Dirichlet boundary conditions [BC]):

$$\underline{u} = \underline{u}_0 \text{ at } \Gamma_u$$

- At Γ_f tractions $\underline{t}_0(t, \underline{X})$ are prescribed (Neumann BC):

$$\underline{\underline{\sigma}} \cdot \underline{n} = \underline{t}_0 \text{ at } \Gamma_f$$

$$\underline{\underline{\sigma}} \cdot \underline{n} = 0 \text{ at } \Gamma_f^0$$



Deformable solid and boundary conditions

Notations:

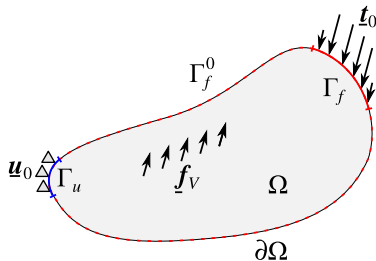
- Consider a solid Ω with boundary $\partial\Omega$
- Boundary is split into Γ_u and Γ_f :
 $\partial\Omega = \Gamma_u \cup \Gamma_f$
- At Γ_u displacements $\underline{u}_0(t, \underline{X})$ are prescribed (Dirichlet boundary conditions [BC]):

$$\underline{u} = \underline{u}_0 \text{ at } \Gamma_u$$

- At Γ_f tractions $\underline{t}_0(t, \underline{X})$ are prescribed (Neumann BC):

$$\underline{\underline{\sigma}} \cdot \underline{n} = \underline{t}_0 \text{ at } \Gamma_f$$

$$\underline{\underline{\sigma}} \cdot \underline{n} = 0 \text{ at } \Gamma_f^0$$



Remarks:

- on the same boundary both BCs can be prescribed if they are orthogonal one to each other, i.e. $\underline{u}_0 \cdot \underline{t}_0 = 0$ (ex.: friction);
- a relationship between these BCs can be prescribed (Robin BC):
 $\underline{u}_0 = \underline{U} - k\underline{t}_0$ (ex.: Winkler's foundation).

Elastic and quasistatic problem set-up

- Equilibrium in absence of inertial forces

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}}_{-V} = 0 \quad (*)$$

- Constitutive relation:

$$\underline{\underline{\sigma}} = {}^4 \underline{\underline{C}} : \underline{\underline{\varepsilon}}$$

- Strain tensor:

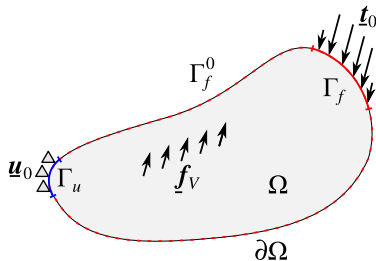
$$\underline{\underline{\varepsilon}} = \frac{1}{2} (\nabla \underline{\underline{u}} + (\nabla \underline{\underline{u}})^T)$$

- Boundary conditions:

$$\underline{\underline{u}} = \underline{\underline{u}}_0 \text{ at } \Gamma_u$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = \underline{\underline{t}}_0 \text{ at } \Gamma_f$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = 0 \text{ at } \Gamma_f^0$$



- **Problem:**

find such field $\underline{\underline{u}}$ in Ω that satisfies equilibrium Eq. (*) and boundary conditions.

Finite Element Method

Main idea

- From continuous to discrete problem

- Split solid into finite elements

$$\Omega \rightarrow \Omega^h \text{ with } \Omega^h = \sum_e \Omega_e^h$$

- All quantities are associated with this discretization:

$$\underline{u} \rightarrow \underline{u}^h, \underline{\sigma} \rightarrow \underline{\sigma}^h, \Gamma_f \rightarrow \Gamma_f^h, \underline{t}_0 \rightarrow \underline{t}_0^h, \dots$$

- Search for \underline{u}^h only in a finite number of points (nodes)

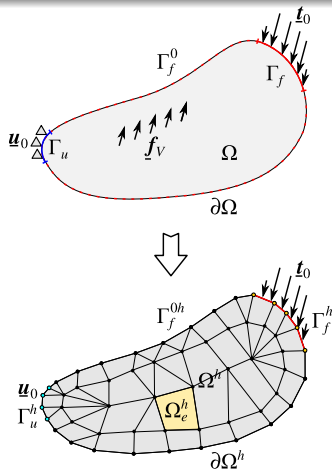
- Interpolate in between (within elements)

- Ensure (1) equilibrium of every element and (2) satisfaction of boundary conditions

$$(1) \quad \nabla \cdot \underline{\sigma}^h + \underline{f}_v^h = 0 \text{ in } \Omega_e^h, \forall e$$

$$(2.a) \quad \underline{\sigma}^h \cdot \underline{n}^h = \underline{t}_0^h \text{ at } \Gamma_f^h$$

$$(2.b) \quad \underline{u}^h = \underline{u}_0^h \text{ at } \Gamma_u^h$$



Main idea

- From continuous to discrete problem

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- All quantities are associated with this discretization:

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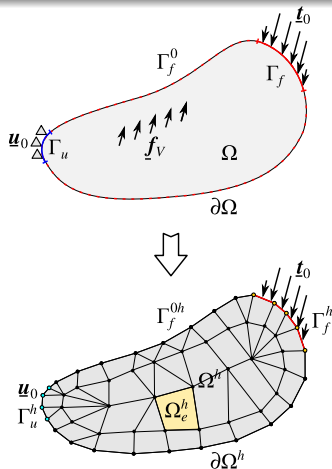
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$$(2.b) \quad \underline{u}^h = \underline{u}_0^h \text{ at } \Gamma_u^h$$



- Existence and uniqueness of the solution \underline{u}_*^h

- When discretization-size tends to zero $h \rightarrow 0$, convergence to the solution of the continuum problem: $\underline{u}_*^h \xrightarrow{h \rightarrow 0} \underline{u}_*$

Standard discrete system

- 1 For any discrete system the quantities of interest $[\mathbf{q}]$ depend on system parameters $[\mathbf{p}]$ and on locally acting external parameters $[\mathbf{e}]$

$$[\mathbf{q}]_i = [\mathbf{q}]_i([\mathbf{p}]_j, [\mathbf{e}]_i)$$

- 2 In the first approximation this dependence is linear

$$q_1 = K_{11}p_1 + K_{12}p_2 + \dots K_{1N}p_N + A_{11}e_1$$

$$q_2 = K_{21}p_1 + K_{22}p_2 + \dots K_{2N}p_N + A_{22}e_2$$

...

$$q_N = K_{N1}p_1 + K_{N2}p_2 + \dots K_{NN}p_N + A_{NN}e_N$$

- 3 In matrix form

$$[\mathbf{q}]_i = [\mathbf{K}]_{ij} [\mathbf{p}]_j + [\mathbf{A}]_{ii} [\mathbf{e}]_i$$

- 4 Assuming that external parameters are of the same nature as quantities of interest ($[\mathbf{A}]_{ij} = [\mathbf{I}]_{ij}$)

$$[\mathbf{q}]_i = [\mathbf{K}]_{ij} [\mathbf{p}]_j + [\mathbf{e}]_i$$

Discrete system in structural mechanics

Main quantities

- Quantities of interest $[\mathbf{q}]$ are, in general, forces $[\mathbf{f}]$
- System parameters $[\mathbf{p}]$ are, in general, displacements $[\mathbf{u}]$
- External parameters $[\mathbf{e}]$ are, in general, external forces $[\mathbf{f}]^{ext}$

Main steps

- 1 Construct *stiffness matrix* and *nodal loads* vector

$$[\mathbf{K}]_{ij}^k, [\mathbf{f}]_i^k, \quad i, j \in 1, NN^k; k \in NE,$$

where NN^k is the number of nodes of k -th element, NE is the number of elements.

- 2 Assemble them into the global stiffness matrix and global load vector

$$[\mathbf{K}]_{ij}, [\mathbf{f}]_i, \quad i, j \in 1, NN,$$

where NN is the total number of nodes.

- 3 Add boundary conditions (for example Dirichlet and Neumann)

$$[\mathbf{f}]_k^{ext}, \quad k \in BC_f; \quad [\mathbf{u}]_l^0, \quad l \in BC_u$$

- 4 Solve linear system of equations

$$[\mathbf{K}]_{ij} [\mathbf{u}]_j = [\mathbf{f}]_i - [\mathbf{f}]_i^{ext} \quad \rightarrow \quad [\mathbf{u}]_{j*}$$

Shape functions

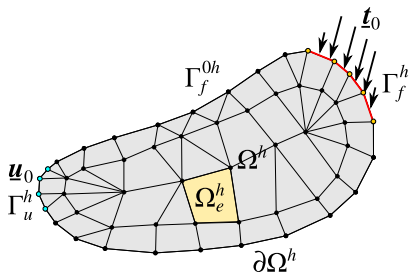
- Displacements are known at nodes: $\underline{u}_i^h, i = 1, 4$
- We need to know them inside the element
- Parametrize the inside with parameters $\{\xi, \eta\} \in [-1, 1]$
- Use *interpolation* or *shape functions* $N_i(\xi, \eta)$ for position \underline{X}

$$\underline{X}^h(\xi, \eta) = \sum_i \underline{X}_i^h N_i(\xi, \eta)$$

and displacement \underline{u} :

$$\underline{u}^h(\xi, \eta) = \sum_i \underline{u}_i^h N_i(\xi, \eta)$$

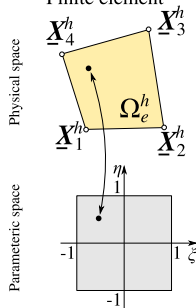
- Remark: Find $\{\xi, \eta\}$ from \underline{X} is not always straightforward (may result in a system of non-linear equations)



Continuum



Finite element



Shape functions

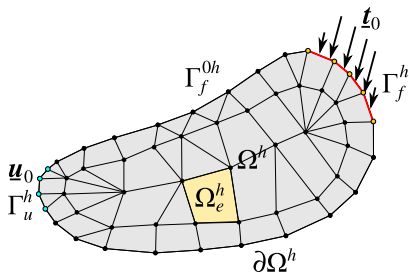
- Displacements are known at nodes: \underline{u}_i^h , $i = 1, 4$
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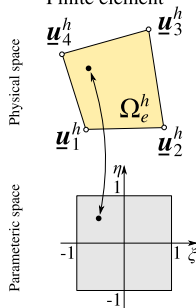
- Remark: Find $\{\xi, \eta\}$ from \underline{X} is not always straightforward (may result in a system of non-linear equations)



Continuum



Finite element



Shape functions II

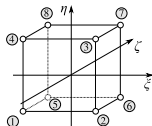
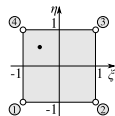
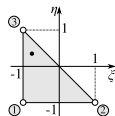
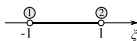
Rules

- Node i has coordinates $\{\xi_i, \eta_i\}$
- Then $N_i(\xi_j, \eta_j) = \delta_{ij}$
- Partition of unity:
$$\forall \xi, \eta, : \sum_i N_i(\xi, \eta) = 1$$

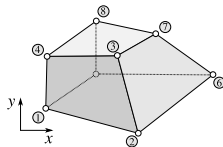
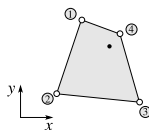
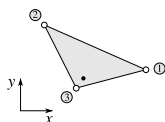
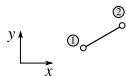
Types

- Linear shape functions
$$\frac{\partial N}{\partial \xi} = \text{const}$$
- Non-linear shape functions
$$\frac{\partial N}{\partial \xi} = f(\xi)$$
- Linear elements vs quadratic elements
- Higher order elements

Parametric space



Physical space



Example: bar element

- Linear shape functions:

$$N_1(\xi) = \frac{1}{2}(1 - \xi)$$

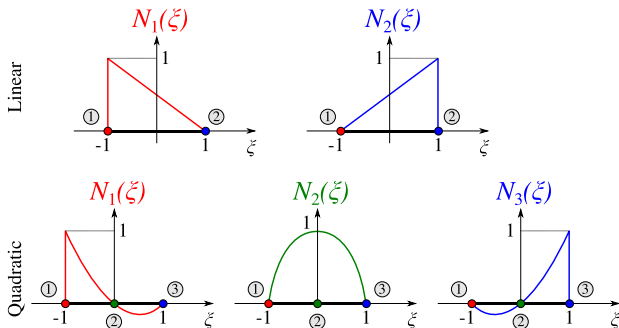
$$N_2(\xi) = \frac{1}{2}(1 + \xi)$$

- Quadratic shape functions:

$$N_1(\xi) = \frac{1}{2}\xi(\xi - 1)$$

$$N_2(\xi) = (1 - \xi^2)$$

$$N_3(\xi) = \frac{1}{2}\xi(1 + \xi)$$



Shape functions: vectors and matrices

- Displacement nodal vectors $\underline{\mathbf{u}}_i = \underline{\mathbf{e}}_x u_i^x + \underline{\mathbf{e}}_y u_i^y$
- Array of nodal coordinates (size $\text{dim} \cdot n$)

$$[\mathbf{X}] = [x_1, y_1, x_2, y_2, \dots, x_n, y_n]_{2n}^T$$

- Array of nodal displacements (size $\text{dim} \cdot n$)

$$[\mathbf{u}] = [u_1^x, u_1^y, u_2^x, u_2^y, \dots, u_n^x, u_n^y]_{2n}^T$$

- Arrays of shape functions (size $\text{dim} \cdot n$)

$$[\mathbf{N}_x] = [N_1, 0, N_2, 0, \dots, N_n, 0]_{2n}^T$$

$$[\mathbf{N}_y] = [0, N_1, 0, N_2, \dots, 0, N_n]_{2n}^T$$

$$[\mathbf{N}] = \begin{bmatrix} N_1 & 0 & N_2 & 0 & \dots & N_n & 0 \\ 0 & N_1 & 0 & N_2 & \dots & 0 & N_n \end{bmatrix}_{2n \times \text{dim}}^T$$

- Then

$$x(\xi, \eta, t) = [\mathbf{N}_x(\xi, \eta)]^T [\mathbf{X}(t)], \quad y(\xi, \eta, t) = [\mathbf{N}_y(\xi, \eta)]^T [\mathbf{X}(t)]$$

$$u^x(\xi, \eta, t) = [\mathbf{N}_x(\xi, \eta)]^T [\mathbf{u}(t)], \quad u^y(\xi, \eta, t) = [\mathbf{N}_y(\xi, \eta)]^T [\mathbf{u}(t)]$$

Gradients and shape functions

- Need to evaluate gradients (spatial derivatives) like $\frac{\partial f}{\partial x}$
- But with shape functions $f = f(\xi, \eta)$
- Then $\frac{\partial f(\xi, \eta)}{\partial x} = \frac{\partial f}{\partial \xi} \frac{\partial \xi}{\partial x} + \frac{\partial f}{\partial \eta} \frac{\partial \eta}{\partial x}$
- However, in general we do not have $\xi = \xi(x, y)$ but rather $x = x(\xi, \eta)$
- Let's do it other way around

$$\begin{bmatrix} \frac{\partial}{\partial \xi} \\ \frac{\partial}{\partial \eta} \end{bmatrix} = \begin{bmatrix} \frac{\partial}{\partial x} \frac{\partial x}{\partial \xi} + \frac{\partial}{\partial y} \frac{\partial y}{\partial \xi} \\ \frac{\partial}{\partial x} \frac{\partial x}{\partial \eta} + \frac{\partial}{\partial y} \frac{\partial y}{\partial \eta} \end{bmatrix} = \begin{bmatrix} \frac{\partial x}{\partial \xi} & \frac{\partial y}{\partial \xi} \\ \frac{\partial x}{\partial \eta} & \frac{\partial y}{\partial \eta} \end{bmatrix} \begin{bmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} \end{bmatrix} = [\mathbf{J}] \begin{bmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} \end{bmatrix}$$

- Matrix $[\mathbf{J}]$ is called Jacobian operator and enables to obtain

$$\begin{bmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} \end{bmatrix} = [\mathbf{J}]^{-1} \begin{bmatrix} \frac{\partial}{\partial \xi} \\ \frac{\partial}{\partial \eta} \end{bmatrix}$$

Jacobian operator

- Jacobian operator or simply Jacobian:

$$[J] = \begin{bmatrix} \frac{\partial x}{\partial \xi} & \frac{\partial y}{\partial \xi} \\ \frac{\partial x}{\partial \eta} & \frac{\partial y}{\partial \eta} \end{bmatrix}$$

- Using $x = [\mathbf{N}_x]^\top [\mathbf{X}]$, $y = [\mathbf{N}_y]^\top [\mathbf{X}]$ we get:

$$[J] = \begin{bmatrix} [\mathbf{N}_{x,\xi}]^\top [\mathbf{X}] & [\mathbf{N}_{y,\xi}]^\top [\mathbf{X}] \\ [\mathbf{N}_{x,\eta}]^\top [\mathbf{X}] & [\mathbf{N}_{y,\eta}]^\top [\mathbf{X}] \end{bmatrix},$$

where $[\mathbf{N}_{x,\xi}] = \left[\frac{\partial N_1}{\partial \xi}, 0, \frac{\partial N_2}{\partial \xi}, 0, \dots, \frac{\partial N_n}{\partial \xi}, 0 \right]^\top$ etc.

- Then the inverse Jacobian is given by:

$$[J]^{-1} = \frac{1}{\Delta} \begin{bmatrix} [\mathbf{N}_{y,\eta}]^\top [\mathbf{X}] & -[\mathbf{N}_{y,\xi}]^\top [\mathbf{X}] \\ -[\mathbf{N}_{x,\eta}]^\top [\mathbf{X}] & [\mathbf{N}_{x,\xi}]^\top [\mathbf{X}] \end{bmatrix},$$

with $\Delta = \det([J]) = [\mathbf{X}]^\top \left([\mathbf{N}_{x,\xi}] [\mathbf{N}_{y,\eta}]^\top - [\mathbf{N}_{y,\xi}] [\mathbf{N}_{x,\eta}]^\top \right) [\mathbf{X}] \neq 0$

Infinitesimal strain in 2D

- Strain tensor: $\underline{\underline{\varepsilon}} = \frac{1}{2} (\nabla \underline{\underline{u}} + (\nabla \underline{\underline{u}})^\top)$ (*)
- Interpolated displacements: $u^x = [\mathbf{N}_x]^\top [\mathbf{u}]$, $u^y = [\mathbf{N}_y]^\top [\mathbf{u}]$
- Displacement gradient:

$$\nabla \underline{\underline{u}} = \underline{\underline{e}}_x \otimes \frac{\partial u^h}{\partial x} + \underline{\underline{e}}_y \otimes \frac{\partial u^h}{\partial y} = \underline{\underline{e}}^x \otimes \underline{\underline{e}}^x \frac{\partial u^x}{\partial x} + \underline{\underline{e}}^x \otimes \underline{\underline{e}}^y \frac{\partial u^y}{\partial x} + \underline{\underline{e}}^y \otimes \underline{\underline{e}}^x \frac{\partial u^x}{\partial y} + \underline{\underline{e}}^y \otimes \underline{\underline{e}}^y \frac{\partial u^y}{\partial y}$$

$$\nabla \underline{\underline{u}} \sim \begin{bmatrix} \frac{\partial u^x}{\partial x} & \frac{\partial u^y}{\partial x} \\ \frac{\partial u^x}{\partial y} & \frac{\partial u^y}{\partial y} \end{bmatrix} = [\mathbf{J}]^{-1} \begin{bmatrix} \frac{\partial}{\partial \xi} \\ \frac{\partial}{\partial \eta} \end{bmatrix} \begin{bmatrix} u^x \\ u^y \end{bmatrix} = [\mathbf{J}]^{-1} \begin{bmatrix} \frac{\partial}{\partial \xi} \\ \frac{\partial}{\partial \eta} \end{bmatrix} \begin{bmatrix} [\mathbf{N}_x]^\top [\mathbf{u}] \\ [\mathbf{N}_y]^\top [\mathbf{u}] \end{bmatrix}$$

- Finally $[\mathbf{E}] = \begin{bmatrix} 1 & 0 & 1 \\ 0 & 1 & 1 \end{bmatrix}^\top [\mathbf{J}]^{-1} \begin{bmatrix} \partial[\mathbf{N}_x]^\top / \partial \xi & \partial[\mathbf{N}_y]^\top / \partial \xi \\ \partial[\mathbf{N}_x]^\top / \partial \eta & \partial[\mathbf{N}_y]^\top / \partial \eta \end{bmatrix} [\mathbf{u}]$

$$\varepsilon_{xx} = \left([\mathbf{J}]_{11}^{-1} [\mathbf{N}_{x,\xi}] + [\mathbf{J}]_{12}^{-1} [\mathbf{N}_{x,\eta}] \right)^\top [\mathbf{u}] = \frac{1}{\Delta} \left([\mathbf{N}_{y,\eta}]^\top [\mathbf{X}] [\mathbf{N}_{x,\xi}] - [\mathbf{N}_{y,\xi}]^\top [\mathbf{X}] [\mathbf{N}_{x,\eta}] \right)^\top [\mathbf{u}]$$

$$\varepsilon_{yy} = \left([\mathbf{J}]_{21}^{-1} [\mathbf{N}_{y,\xi}] + [\mathbf{J}]_{22}^{-1} [\mathbf{N}_{y,\eta}] \right)^\top [\mathbf{u}] = \frac{1}{\Delta} \left(-[\mathbf{N}_{x,\eta}]^\top [\mathbf{X}] [\mathbf{N}_{y,\xi}] + [\mathbf{N}_{x,\xi}]^\top [\mathbf{X}] [\mathbf{N}_{y,\eta}] \right)^\top [\mathbf{u}]$$

$$\varepsilon_{xy} = \frac{1}{2} \left(\frac{\partial u^x}{\partial y} + \frac{\partial u^y}{\partial x} \right) = \frac{1}{2} \left([\mathbf{J}]_{11}^{-1} [\mathbf{N}_{y,\xi}] + [\mathbf{J}]_{12}^{-1} [\mathbf{N}_{y,\eta}] + [\mathbf{J}]_{21}^{-1} [\mathbf{N}_{x,\xi}] + [\mathbf{J}]_{22}^{-1} [\mathbf{N}_{x,\eta}] \right)^\top [\mathbf{u}]$$

Infinitesimal strain in 2D

- Strain tensor: $\underline{\underline{\varepsilon}} = \frac{1}{2} (\nabla \underline{\mathbf{u}} + (\nabla \underline{\mathbf{u}})^\top)$ (*)
- Interpolated displacements: $u^x = [\mathbf{N}_x]^\top [\mathbf{u}]$, $u^y = [\mathbf{N}_y]^\top [\mathbf{u}]$
- Displacement gradient:

$$\nabla \underline{\mathbf{u}} = \underline{\mathbf{e}}_x \otimes \frac{\partial u^h}{\partial x} + \underline{\mathbf{e}}_y \otimes \frac{\partial u^h}{\partial y} = \underline{\mathbf{e}}^x \otimes \underline{\mathbf{e}}^x \frac{\partial u^x}{\partial x} + \underline{\mathbf{e}}^x \otimes \underline{\mathbf{e}}^y \frac{\partial u^y}{\partial x} + \underline{\mathbf{e}}^y \otimes \underline{\mathbf{e}}^x \frac{\partial u^x}{\partial y} + \underline{\mathbf{e}}^y \otimes \underline{\mathbf{e}}^y \frac{\partial u^y}{\partial y}$$

$$\nabla \underline{\mathbf{u}} \sim \begin{bmatrix} \frac{\partial u^x}{\partial x} & \frac{\partial u^y}{\partial x} \\ \frac{\partial u^x}{\partial y} & \frac{\partial u^y}{\partial y} \end{bmatrix} = [\mathbf{J}]^{-1} \begin{bmatrix} \frac{\partial}{\partial \xi} \\ \frac{\partial}{\partial \eta} \end{bmatrix} \begin{bmatrix} u^x \\ u^y \end{bmatrix} = [\mathbf{J}]^{-1} \begin{bmatrix} \frac{\partial}{\partial \xi} \\ \frac{\partial}{\partial \eta} \end{bmatrix} \begin{bmatrix} [\mathbf{N}_x]^\top [\mathbf{u}] \\ [\mathbf{N}_y]^\top [\mathbf{u}] \end{bmatrix}$$

- Finally $[\mathbf{E}] = \begin{bmatrix} 1 & 0 & 1 \\ 0 & 1 & 1 \end{bmatrix}^\top [\mathbf{J}]^{-1} \begin{bmatrix} \partial[\mathbf{N}_x]^\top / \partial \xi & \partial[\mathbf{N}_y]^\top / \partial \xi \\ \partial[\mathbf{N}_x]^\top / \partial \eta & \partial[\mathbf{N}_y]^\top / \partial \eta \end{bmatrix} [\mathbf{u}]$

$$\varepsilon_{xx} = \left([\mathbf{J}]_{11}^{-1} [\mathbf{N}_{x,\xi}] + [\mathbf{J}]_{12}^{-1} [\mathbf{N}_{x,\eta}] \right)^\top [\mathbf{u}] = \frac{1}{\Delta} \left([\mathbf{N}_{y,\eta}]^\top [\mathbf{X}] [\mathbf{N}_{x,\xi}] - [\mathbf{N}_{y,\xi}]^\top [\mathbf{X}] [\mathbf{N}_{x,\eta}] \right)^\top [\mathbf{u}]$$

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$$\varepsilon_{xy} = \frac{1}{2\Delta} \left([\mathbf{N}_{y,\eta}]^\top [\mathbf{X}] [\mathbf{N}_{y,\xi}] - [\mathbf{N}_{y,\xi}]^\top [\mathbf{X}] [\mathbf{N}_{y,\eta}] - [\mathbf{N}_{x,\eta}]^\top [\mathbf{X}] [\mathbf{N}_{x,\xi}] + [\mathbf{N}_{x,\xi}]^\top [\mathbf{X}] [\mathbf{N}_{x,\eta}] \right)^\top [\mathbf{u}]$$

Infinitesimal strain in 2D in matrix form

- Strain tensor: $\underline{\underline{\varepsilon}} = \frac{1}{2} (\nabla \underline{\mathbf{u}} + (\nabla \underline{\mathbf{u}})^T)$ (*)
- Represent it as an array (Voigt notations):

$$\underline{\underline{\varepsilon}} \Rightarrow [\mathbf{E}] = [\varepsilon_{xx}, \varepsilon_{yy}, \gamma_{xy}]^T, \quad \gamma_{xy} = 2\varepsilon_{xy}$$

- Then

$$[\mathbf{E}]_3 = [\mathbf{B}]_{3 \times 2n}^T [\mathbf{u}]_{2n}$$

- With $[\mathbf{B}]$ given by:

$$[\mathbf{B}]^T = \frac{1}{\Delta} \begin{bmatrix} ([\mathbf{N}_{y,\eta}]^T [\mathbf{X}] [\mathbf{N}_{x,\xi}] - [\mathbf{N}_{y,\xi}]^T [\mathbf{X}] [\mathbf{N}_{x,\eta}])^T \\ (-[\mathbf{N}_{x,\eta}]^T [\mathbf{X}] [\mathbf{N}_{y,\xi}] + [\mathbf{N}_{x,\xi}]^T [\mathbf{X}] [\mathbf{N}_{y,\eta}])^T \\ (([\mathbf{N}_{y,\eta}]^T [\mathbf{X}] [\mathbf{N}_{y,\xi}] - [\mathbf{N}_{y,\xi}]^T [\mathbf{X}] [\mathbf{N}_{y,\eta}] - [\mathbf{N}_{x,\eta}]^T [\mathbf{X}] [\mathbf{N}_{x,\xi}] + [\mathbf{N}_{x,\xi}]^T [\mathbf{X}] [\mathbf{N}_{x,\eta}])^T \end{bmatrix}_{3 \times 2n}$$

Infinitesimal strain in 2D: example

- Consider a linear triangular element with shape functions:

$$N_1 = -\frac{1}{2}(\xi + \eta), \quad N_2 = \frac{1}{2}(1 + \xi), \quad N_3 = \frac{1}{2}(1 + \eta)$$

- Their derivatives are given by:

$$N_{1,\xi} = -1/2, \quad N_{2,\xi} = 1/2, \quad N_{3,\xi} = 0$$

$$N_{1,\eta} = -1/2, \quad N_{2,\eta} = 0, \quad N_{3,\eta} = 1/2$$

$$\Delta = \frac{1}{4}((x_2 - x_1)(y_3 - y_1) - (y_2 - y_1)(x_3 - x_1))^*$$

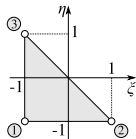
- Then

$$\varepsilon_{xx} = \frac{1}{4\Delta} \left[(y_3 - y_1)(u_2^x - u_1^x) - (y_2 - y_1)(u_3^x - u_1^x) \right]$$

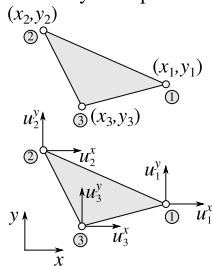
$$\varepsilon_{yy} = \frac{1}{4\Delta} \left[(x_2 - x_1)(u_3^y - u_1^y) - (x_3 - x_1)(u_2^y - u_1^y) \right]$$

$$\gamma_{xy} = \frac{1}{4\Delta} \left[(y_3 - y_1)(u_2^y - u_1^y) - (y_2 - y_1)(u_3^y - u_1^y) + (x_2 - x_1)(u_3^x - u_1^x) - (x_3 - x_1)(u_2^x - u_1^x) \right]$$

Parametric space



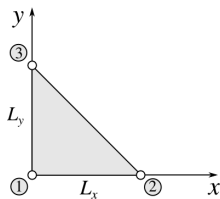
Physical space



*Half of the area of the triangle.

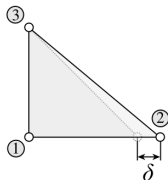
Infinitesimal strain in 2D: example II

- Rectangular triangle $x_1 = x_3$, $y_1 = y_2$, $\Delta = L_x L_y / 4$
- Case 1: pure tension/compression along OX axis $u_3^y = u_1^y$, $u_2^y = u_1^y$, $u_3^x = u_1^x$
Ex.: $u_2^x = \delta$: $\varepsilon_{xx} = \frac{1}{4\Delta}(y_3 - y_1)(u_2^x - u_1^x) = \delta/L_x$, $\varepsilon_{yy} = \gamma_{xy} = 0$



Reference configuration

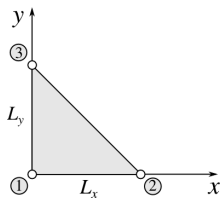
Case 1



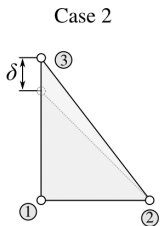
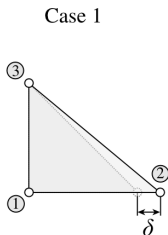
Current configuration

Infinitesimal strain in 2D: example II

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- Case 2: pure tension/compression along OY iaoi $u_2^x = u_1^x$, $u_2^y = u_1^y$, $u_3^x = u_1^x$
Ex.: $u_3^y = \delta$: $\epsilon_{yy} = \frac{1}{4\Delta}(x_2 - x_1)(u_3^y - u_1^y) = \delta/L_y$, $\epsilon_{xx} = \gamma_{xy} = 0$



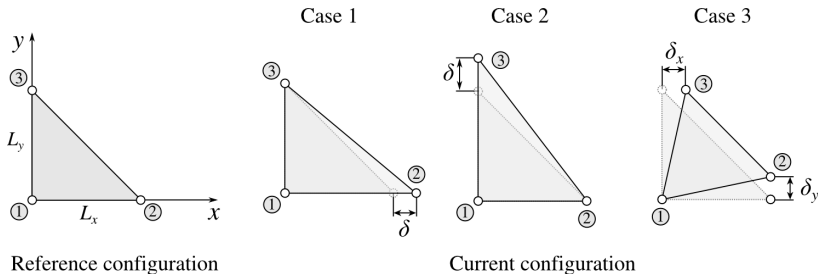
Reference configuration



Current configuration

Infinitesimal strain in 2D: example II

- Rectangular triangle $x_1 = x_3$, $y_1 = y_2$, $\Delta = L_x L_y / 4$
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Ex.: $u_2^x = \delta$: $\epsilon_{xx} = \frac{1}{4\Delta}(y_3 - y_1)(u_2^x - u_1^x) = \delta/L_x$, $\epsilon_{yy} = \gamma_{xy} = 0$
- Case 2: pure tension/compression along OY iaoi $u_2^x = u_1^x$, $u_2^y = u_1^y$, $u_3^x = u_1^x$
Ex.: $u_3^y = \delta$: $\epsilon_{yy} = \frac{1}{4\Delta}(x_2 - x_1)(u_3^y - u_1^y) = \delta/L_y$, $\epsilon_{xx} = \gamma_{xy} = 0$
- Case 3: pure shear in XY iaoi $u_2^x = u_1^x$, $u_3^y = u_1^y$
Ex.: $u_2^y = \delta_y$, $u_3^x = \delta_x$:
$$\gamma_{xy} = \frac{1}{4\Delta} \left((y_3 - y_1)(u_2^y - u_1^y) + (x_2 - x_1)(u_3^x - u_1^x) \right) = \frac{\delta_y}{L_x} + \frac{\delta_x}{L_y}, \quad \epsilon_{xx} = \epsilon_{yy} = 0$$



- In linear elasticity:

$$\underline{\underline{\sigma}} = {}^4\underline{\underline{C}} : (\underline{\underline{\varepsilon}} - \underline{\underline{\varepsilon}}_0) + \underline{\underline{\sigma}}_0$$

- Residual stress field $\underline{\underline{\sigma}}_0$

- Initial strain field $\underline{\underline{\varepsilon}}_0$

- In self equilibrated system: $\underline{\underline{\sigma}}_0 = {}^4\underline{\underline{C}} : \underline{\underline{\varepsilon}}_0$ resulting in

$$\underline{\underline{\sigma}} = {}^4\underline{\underline{C}} : (\underline{\underline{\varepsilon}} - \underline{\underline{\varepsilon}}_{th})$$

- With thermal strain field $\underline{\underline{\varepsilon}}_{th}$:

$$\underline{\underline{\varepsilon}}_{th} = \alpha(T - T_0)\underline{\underline{I}}$$

where α is the coefficient of thermal expansion (CTE), T and T_0 are the current and reference temperature fields, respectively.

Stress: 2D isotropic elasticity

- Recall stress/strain relationship:

$$\underline{\underline{\sigma}} = \frac{\nu E}{(1+\nu)(1-2\nu)} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{I}} + \frac{E}{1+\nu} \underline{\underline{\epsilon}}$$

- Stress (in Voigt notations): $\underline{\underline{\sigma}} \Rightarrow [\mathbf{S}] = [\sigma_{xx}, \sigma_{yy}, \sigma_{xy}]^T$
- In plane stress $\sigma_{zz} = 0, \epsilon_{zz} = \frac{\nu}{\nu-1}(\epsilon_{xx} + \epsilon_{yy})$
- In plain strain $\sigma_{zz} = \nu(\sigma_{xx} + \sigma_{yy}), \epsilon_{zz} = 0$
- Stress/strain relationship: $[\mathbf{S}]_i = [\mathbf{D}]_{ij} [\mathbf{E}]_j$
- Matrix $[\mathbf{D}]$ in plane strain $\epsilon_{zz} = \epsilon_{xz} = \epsilon_{yz} = 0$:

$$[\mathbf{D}]_{ij} = \frac{E}{(1+\nu)(1-2\nu)} \begin{bmatrix} 1-\nu & \nu & 0 \\ \nu & 1-\nu & 0 \\ 0 & 0 & (1-2\nu)/2^* \end{bmatrix}$$

- Matrix $[\mathbf{D}]$ in plane stress $\sigma_{zz} = \sigma_{xz} = \sigma_{yz} = 0, \text{tr}(\underline{\underline{\epsilon}}) = \frac{1-2\nu}{1-\nu}(\epsilon_{xx} + \epsilon_{yy})$:

$$[\mathbf{D}]_{ij} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & (1-\nu)/2^* \end{bmatrix}$$

*Factor 1/2 appears because γ_{xy} was inserted in $[\mathbf{E}]$ instead of ϵ_{xy} .

Voigt notations in 3D case

- Stress tensor: $\underline{\underline{\sigma}} \rightarrow [\mathbf{S}] = [\sigma_{xx}, \sigma_{yy}, \sigma_{zz}, \sigma_{xy}, \sigma_{yz}, \sigma_{xz}]^T$
- Strain tensor: $\underline{\underline{\varepsilon}} \rightarrow [\mathbf{E}] = [\varepsilon_{xx}, \varepsilon_{yy}, \varepsilon_{zz}, \gamma_{xy}, \gamma_{yz}, \gamma_{xz}]^T$
- Hooke's law: $[\mathbf{S}] = [\mathbf{D}][\mathbf{E}]$
- Isotropic elasticity (two constants E, ν):

$$[\mathbf{D}]_{ij} = \frac{E}{(1+\nu)(1-2\nu)} \begin{bmatrix} 1-\nu & \nu & \nu & 0 & 0 & 0 \\ \nu & 1-\nu & \nu & 0 & 0 & 0 \\ \nu & \nu & 1-\nu & 0 & 0 & 0 \\ 0 & 0 & 0 & (1-2\nu)/2 & 0 & 0 \\ 0 & 0 & 0 & 0 & (1-2\nu)/2 & 0 \\ 0 & 0 & 0 & 0 & 0 & (1-2\nu)/2 \end{bmatrix}$$

- Cubic elasticity (3 constants E, ν, μ):

$$[\mathbf{D}]_{ij} = \begin{bmatrix} C_{11} & C_{12} & C_{12} & 0 & 0 & 0 \\ C_{12} & C_{11} & C_{12} & 0 & 0 & 0 \\ C_{12} & C_{12} & C_{11} & 0 & 0 & 0 \\ 0 & 0 & 0 & C_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & C_{44} & 0 \\ 0 & 0 & 0 & 0 & 0 & C_{44} \end{bmatrix}$$

Stress: general case II

Voigt notations in 3D case

- Transversely isotropic elasticity (5 constants $E_1, E_2, \nu_1, \nu_2, \mu_1$):

$$[\mathbf{D}]_{ij} = \begin{bmatrix} C_{11} & C_{12} & C_{13} & 0 & 0 & 0 \\ C_{12} & C_{11} & C_{13} & 0 & 0 & 0 \\ C_{13} & C_{13} & C_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & C_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & C_{44} & 0 \\ 0 & 0 & 0 & 0 & 0 & (C_{11} - C_{12})/2 \end{bmatrix}$$

- Orthotropic elasticity (9 constants $E_{xx}, E_{yy}, E_{zz}, \nu_{xy}, \nu_{yz}, \nu_{xz}, \mu_{xy}, \mu_{yz}, \mu_{xz}$):

$$[\mathbf{D}]_{ij} = \begin{bmatrix} C_{11} & C_{12} & C_{13} & 0 & 0 & 0 \\ C_{12} & C_{22} & C_{23} & 0 & 0 & 0 \\ C_{13} & C_{23} & C_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & C_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & C_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & C_{66} \end{bmatrix}$$

Strain/Stress: spherical part

Spherical part of a tensor = $\frac{1}{3}\text{tr}(\underline{\underline{A}})\underline{\underline{I}}$

- If the strain tensor can be presented as $\underline{\underline{\varepsilon}} = \frac{1}{3}\text{tr}(\underline{\underline{\varepsilon}})\underline{\underline{I}}$,
then only volume change happens at this location $\Delta V/V_0 = \text{tr}(\underline{\underline{\varepsilon}})$

$$\underline{\underline{\varepsilon}} \sim \begin{bmatrix} \varepsilon & 0 & 0 \\ 0 & \varepsilon & 0 \\ 0 & 0 & \varepsilon \end{bmatrix}$$

- If the stress tensor can be presented as $\underline{\underline{\sigma}} = \frac{1}{3}\text{tr}(\underline{\underline{\sigma}})\underline{\underline{I}}$,
then the stress state is pure hydrostatic compression under pressure
 $p = -\text{tr}(\underline{\underline{\sigma}})/3$

$$\underline{\underline{\sigma}} \sim \begin{bmatrix} -p & 0 & 0 \\ 0 & -p & 0 \\ 0 & 0 & -p \end{bmatrix}$$

Strain/Stress: deviatoric part

Deviatoric part of a tensor = $\underline{\underline{A}} - \frac{1}{3}\text{tr}(\underline{\underline{A}})\underline{\underline{I}}$

- If the strain tensor does not have spherical part $\underline{\underline{\varepsilon}} = \underline{\underline{\varepsilon}} - \frac{1}{3}\text{tr}(\underline{\underline{\varepsilon}})\underline{\underline{I}}$, then no volume change happens at this location $\Delta V/V_0 = 0$ only the shape changes, Ex.:

$$\underline{\underline{\varepsilon}} \sim \begin{bmatrix} \varepsilon & 0 & 0 \\ 0 & -0.5\varepsilon & 0 \\ 0 & 0 & -0.5\varepsilon \end{bmatrix}, \quad \underline{\underline{\varepsilon}} \sim \begin{bmatrix} 0 & \varepsilon & 0 \\ \varepsilon & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

- If the stress tensor is presented only by deviatoric part $\underline{\underline{\sigma}} = \underline{\underline{\sigma}} - \frac{1}{3}\text{tr}(\underline{\underline{\sigma}})\underline{\underline{I}}$, then the stress state is pure shear:

$$\underline{\underline{\sigma}} \sim \begin{bmatrix} -\sigma & 0 & 0 \\ 0 & 2\sigma & 0 \\ 0 & 0 & -\sigma \end{bmatrix}, \quad \underline{\underline{\sigma}} \sim \begin{bmatrix} 0 & \sigma_{xy} & \sigma_{xz} \\ \sigma_{xy} & 0 & 0 \\ \sigma_{xz} & 0 & 0 \end{bmatrix}$$

- In general both parts are present: $\underline{\underline{\varepsilon}} = \underline{\underline{e}} + \text{tr}(\underline{\underline{\varepsilon}})\underline{\underline{I}}/3$, $\underline{\underline{\sigma}} = \underline{\underline{s}} + \text{tr}(\underline{\underline{\sigma}})\underline{\underline{I}}/3$

Strain/Stress: elastic relationships

- Recall: $\underline{\underline{\epsilon}} = \underline{\underline{e}} + \frac{\Delta V}{3V} \underline{\underline{I}}$, $\underline{\underline{\sigma}} = \underline{\underline{s}} - p \underline{\underline{I}}$
- For deviatoric part in linear isotropic elasticity

$$\underline{\underline{s}} = \frac{E}{1+\nu} \underline{\underline{e}}, \quad \underline{\underline{e}} = 2\mu \underline{\underline{e}}$$

where $\mu = \frac{E}{2(1+\nu)}$ is called *shear modulus*.

- For spherical parts

$$\text{tr}(\underline{\underline{\epsilon}}) = \frac{1-2\nu}{E} \text{tr}(\sigma) = -\frac{3(1-2\nu)}{E} p$$

then

$$-\frac{1}{V} \frac{dV}{dp} = \frac{3(1-2\nu)}{E} \Leftrightarrow -V \frac{dp}{dV} = \frac{E}{3(1-2\nu)} = K$$

where $K = \frac{E}{3(1-2\nu)}$ is called *bulk modulus*.

Stress and reactions: element's equilibrium II

- Work of nodal forces on *virtual* nodal displacements = $\frac{1}{2} \underline{f}_{-i} \cdot \delta \underline{u}_i$
- Work density of distributed volumetric forces = $\frac{1}{2} \underline{f}_{-v} \cdot \delta \underline{u}_v$
- Corresponding density of elastic energy = $\frac{1}{2} \underline{\sigma} : \delta \underline{\epsilon}$

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- Corresponding density of elastic energy = $\frac{1}{2} \underline{\underline{\sigma}} : \delta \underline{\underline{\epsilon}}$
- Stored elastic energy equals this work:

$$\int_{V^e} \underline{\underline{\sigma}} : \underline{\underline{\epsilon}} dV = \sum_i \underline{f}_{-i} \cdot \underline{u}_i + \int V^e \underline{f}_{-V} \cdot \delta \underline{u} dV$$

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- Elastic stress $\underline{\underline{\sigma}} = {}^4 \underline{\underline{C}} : (\underline{\underline{\epsilon}} - \underline{\underline{\epsilon}}_{th}) \Rightarrow [\mathbf{S}] = [\mathbf{D}] ([\mathbf{E}] - [\mathbf{E}_{th}])$
- Strain $\underline{\underline{\epsilon}} \sim [\mathbf{E}] = [\mathbf{B}]^T [\mathbf{u}]$, vol. force density $\underline{f}_{-V} \sim [\mathbf{f}_V] = [f_v^x, f_v^y, f_v^z]^T$, volumetric virt. displacement $\delta \underline{u}_V \sim [\mathbf{N}]^T \delta [\mathbf{u}]$:

$$\int_{V^e} \{ ([\mathbf{D}] ([\mathbf{E}] - [\mathbf{E}_{th}]))^T \delta [\mathbf{E}] - [\mathbf{f}_V]^T [\mathbf{N}_i]^T \delta [\mathbf{u}] \} dV = [\mathbf{f}]^T \delta [\mathbf{u}]$$

Stress and reactions: element's equilibrium II

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$$\int_{V^e} \{ ([\underline{D}] ([\underline{E}] - [\underline{E}_{th}]))^T \delta [\underline{E}] - [\underline{f}_v]^T [\underline{N}_i]^T \delta [\underline{u}] \} dV = [\underline{f}]^T \delta [\underline{u}]$$
$$[\underline{u}] \left[\int_{V^e} [\underline{B}] [\underline{D}] [\underline{B}]^T dV \right] \delta [\underline{u}] - \left[\int_{V^e} ([\underline{f}_v]^T [\underline{N}_i]^T + [\underline{E}_{th}]^T [\underline{D}] [\underline{B}]^T) dV \right] \delta [\underline{u}] = [\underline{f}]^T \delta [\underline{u}]$$

Stress and reactions: element's equilibrium II

- Balance of virtual work for a single element:

$$[\mathbf{u}] \left[\int_{V^e} [\mathbf{B}] [\mathbf{D}] [\mathbf{B}]^T dV \right] \delta[\mathbf{u}] - \left[\int_{V^e} ([\mathbf{f}_v]^T [\mathbf{N}_i]^T + [\mathbf{E}_{th}]^T [\mathbf{D}] [\mathbf{B}]^T) dV \right] \delta[\mathbf{u}] = [\mathbf{f}]^T \delta[\mathbf{u}]$$

- For arbitrary virtual displacements $\delta[\mathbf{u}]$:

$$\underbrace{\left[\int_{V^e} [\mathbf{B}]^T [\mathbf{D}] [\mathbf{B}] dV \right]}_{[\mathbf{K}^e]} [\mathbf{u}] + \underbrace{\left[\int_{V^e} (-[\mathbf{f}_v]^T [\mathbf{N}_i] - [\mathbf{B}] [\mathbf{D}] [\mathbf{E}_{th}]) dV \right]}_{[\mathbf{f}_{int}^e]} = \underbrace{[\mathbf{f}]}_{[\mathbf{f}_{ext}^e]}$$

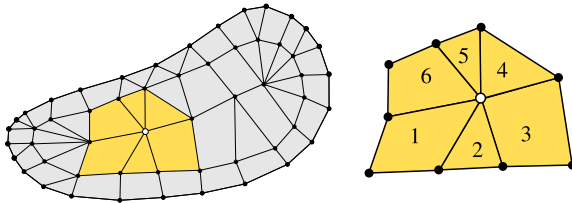
- System of equations linking displacements and reactions:

$$\boxed{[\mathbf{K}^e] [\mathbf{u}^e] + [\mathbf{f}_{int}^e] = [\mathbf{f}_{ext}^e]}$$

- At every internal node the total force should be zero:

$$\sum_e [\mathbf{f}_{\text{ext}}^e] = 0$$

summation over all elements e attached to this node.



- Summation over all nodes gives:

$$[\mathbf{K}][\mathbf{u}] + [\mathbf{f}_{\text{int}}] = 0$$

Dirichlet boundary conditions

Dirichlet BC

- Use penalty method to enforce prescribed displacements: array $[\mathbf{u}_0] = [0 \dots 0 u_{i0} 0 \dots 0 u_{j0} 0]$
- Diagonal selection matrix $[\mathbf{I}^s]$ with ones at prescribed degrees of freedom (DOFs):

$$[\mathbf{I}^s] = \begin{bmatrix} 0 & \dots & 0 & \underbrace{0}_i & 0 & \dots & 0 & \underbrace{0}_j & 0 \\ \vdots & & \vdots & \vdots & \vdots & & \vdots & \vdots & \vdots \\ 0 & \dots & 0 & 0 & 0 & \dots & 0 & 0 & 0 \\ 0 & \dots & 0 & \mathbf{1} & 0 & \dots & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & 0 & \dots & 0 & 0 & 0 \\ \vdots & & \vdots & \vdots & \vdots & & \vdots & \vdots & \vdots \\ 0 & \dots & 0 & 0 & 0 & \dots & 0 & 0 & 0 \\ 0 & \dots & 0 & 0 & 0 & \dots & 0 & \mathbf{1} & 0 \\ 0 & \dots & 0 & 0 & 0 & \dots & 0 & 0 & 0 \end{bmatrix} \begin{matrix} \\ \\ \\ \} i \\ \\ \\ \\ \} j \end{matrix}$$

- Then the system is changed to

$$([\mathbf{K}] + \epsilon [\mathbf{I}^s]) [\mathbf{u}] = ([\mathbf{I}] - [\mathbf{I}^s]) ([\mathbf{f}_{\text{ext}}] - [\mathbf{f}_{\text{int}}]) + \epsilon [\mathbf{u}_0]$$

where ϵ is the penalty coefficient such that $\epsilon \gg \max(K_{ij})$, and $[\mathbf{I}]$ is the identity matrix.

Neumann boundary conditions

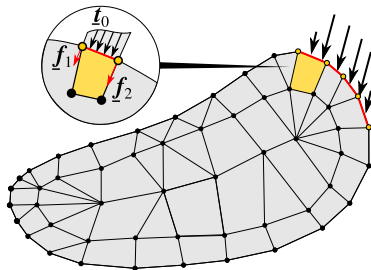
Neumann BC

- Surface traction \underline{t}_0 at Γ_f
- Virtual work of surface traction over one element:

$$\int_{\Gamma_f^e} \underline{t}_0 \cdot \delta \underline{u} d\Gamma = \underline{f}_{-ext}^i \cdot \delta \underline{u}_i^e$$

- Then

$$[\underline{f}_{ext}^i] = \int_{\Gamma_f^e} [\underline{t}_0]^T [\mathbf{N}]^T d\Gamma$$



Discrete system of equations

- Balance of virtual work for the whole body:

$$\underbrace{\left[\int_V [\mathbf{B}]^T [\mathbf{D}] [\mathbf{B}] dV \right]}_{[\mathbf{K}]} [\mathbf{u}] = \underbrace{\int_{\Gamma_f} [\mathbf{t}_0]^T [\mathbf{N}]^T d\Gamma}_{[\mathbf{f}_{\text{ext}}]} + \underbrace{\left[\int_V \left([\mathbf{f}_v]^T [\mathbf{N}_i] + [\mathbf{B}] [\mathbf{D}] [\mathbf{E}_{\text{th}}] \right) dV \right]}_{-[\mathbf{f}_{\text{int}}]}$$

- System of equations linking displacements and reactions:

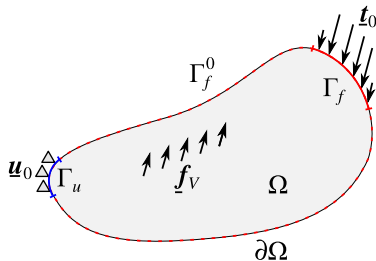
$$\boxed{[\mathbf{K}] [\mathbf{u}] = [\mathbf{f}_{\text{ext}}] - [\mathbf{f}_{\text{int}}]}$$

- Stiffness matrix $[\mathbf{K}]$
- Vector of degrees of freedom (DOFs) $[\mathbf{u}]$
- Right hand term (vector of forces) $[\mathbf{f}_{\text{ext}}] - [\mathbf{f}_{\text{int}}]$

Different approach: virtual work formulation I

- Arbitrary virtual displacements $\delta \underline{u}$
- Strong form: $\nabla \cdot \underline{\underline{\sigma}} + \underline{f}_{-V} = 0 + \text{BCs}$
- Take a product with virtual displacements and integrate over Ω :

$$\int_{\Omega} \left(\nabla \cdot \underline{\underline{\sigma}} \cdot \delta \underline{u} + \underline{f}_{-V} \cdot \delta \underline{u} \right) dV = 0$$



- Replacement: $\nabla \cdot \underline{\underline{\sigma}} \cdot \delta \underline{u} = \nabla \cdot (\underline{\underline{\sigma}} \cdot \delta \underline{u}) - \underline{\underline{\sigma}} : \nabla \delta \underline{u}$
- Following Gauss-Ostrogradsky theorem: $\int_V \nabla \cdot (\bullet) dV = \int_S \underline{n} \cdot (\bullet) dS$
- So

$$\int_{\partial\Omega} \underline{n} \cdot \underline{\underline{\sigma}} \cdot \delta \underline{u} dS + \int_{\Omega} \left(\underline{f}_{-V} \cdot \delta \underline{u} - \underline{\underline{\sigma}} : \delta \underline{\underline{\varepsilon}} \right) dV = 0$$

Different approach: virtual work formulation II

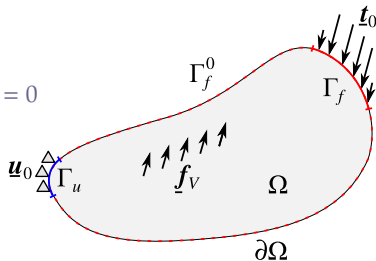
continue...

- Weak form

$$\int_{\partial\Omega} \underline{n} \cdot \underline{\sigma} \cdot \delta \underline{u} dS + \int_{\Omega} (\underline{f}_{-V} \cdot \delta \underline{u} - \underline{\sigma} : \delta \underline{\underline{\varepsilon}}) dV = 0$$

- Non-trivial Neumann boundary conditions at Γ_f

$$\int_{\Omega} \underline{\sigma} : \delta \underline{\underline{\varepsilon}} dV = \int_{\Gamma_f} \underline{t}_0 \cdot \delta \underline{u} dS + \int_{\Omega} \underline{f}_{-V} \cdot \delta \underline{u} dV$$



- Remark I:** in the strong form \underline{u} should be C^2 -smooth, in the weak form \underline{u} should be only square-integrable as well as its first derivative, thus $\underline{u} \in \mathbb{H}^1$, i.e. from Sobolev's functional space of the first order. In addition $\underline{u} = \underline{u}_0$ at Γ_u
- Remark II:** for linear elasticity, the stress tensor* $\underline{\sigma} = {}^4\underline{\underline{C}} : (\underline{\underline{\varepsilon}} - \underline{\underline{\varepsilon}}_{th})$

$$\int_{\Omega} \underline{\underline{\varepsilon}} : {}^4\underline{\underline{C}} : \delta \underline{\underline{\varepsilon}} dV = \int_{\Gamma_f} \underline{t}_0 \cdot \delta \underline{u} dS + \int_{\Omega} (\underline{f}_{-V} + {}^4\underline{\underline{C}} : \underline{\underline{\varepsilon}}_{th}) \cdot \delta \underline{u} dV$$

Different approach II: potential energy

Remark III:

- If the system remains linear (boundary conditions, linear elasticity)
- The principle of virtual work is equivalent to the minimum of the total potential energy
- {Potential energy} = {Internal energy} - {Work of all forces}

$$\Pi(\underline{u}, \underline{t}_0, \underline{u}_0) = \frac{1}{2} \int_{\Omega} \underline{\underline{\sigma}} : \underline{\underline{\varepsilon}} dV - \int_{\Gamma_f} \underline{t}_0 \cdot \underline{u} d\Gamma - \int_{\Omega} \underline{f}_{-V} \cdot \underline{u} dV$$

- Stationary point of the total potential energy $\frac{\partial \Pi}{\partial \underline{u}} = 0$ for given loads $\underline{t}_0, \underline{u}_0$:

$$\frac{\partial \Pi}{\partial \underline{u}} = \int_{\Omega} \underline{\underline{\varepsilon}} : {}^4 \underline{\underline{C}} : \frac{\partial \underline{\underline{\varepsilon}}}{\partial \underline{u}} dV - \int_{\Gamma_f} \underline{t}_0 d\Gamma - \int_{\Omega} \underline{f}_{-V} dV = 0$$

- The same equation

Evaluation of the integrals

- Weak form (recall):

$$\underbrace{\left[\int_V [\mathbf{B}]^T [\mathbf{D}] [\mathbf{B}] dV \right]}_{[\mathbf{K}]} [\mathbf{u}] = \underbrace{\int_{\Gamma_f} [\mathbf{t}_0]^T [\mathbf{N}]^T d\Gamma}_{[\mathbf{f}_{\text{ext}}]} + \underbrace{\left[\int_V ([\mathbf{f}_v]^T [\mathbf{N}_i] + [\mathbf{B}] [\mathbf{D}] [\mathbf{E}_{\text{th}}]) dV \right]}_{-[\mathbf{f}_{\text{int}}]}$$

- Exact integration: $\int_a^b f(x) dx = F(b) - F(a)$ (not always possible)
- Approximate integration (trapezoidal rule, Simpson's rule)
- Gauss quadrature: $\int_a^b f(x) dx \approx \sum_{i=1}^{N_{GP}} w_i f(x_i)$
- Gauss points x_i with $i = 1, N_{GP}$
- Integration is exact for polynomials of order $2N_{GP} - 1$
- Tabulated data for x_i, w_i (1D, 2D, 3D integration)

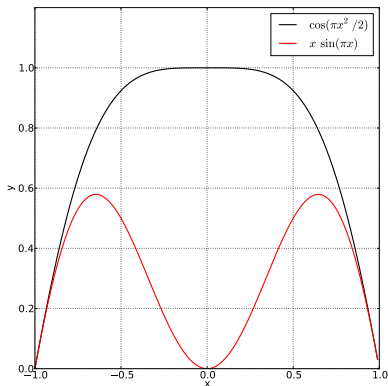
Evaluation of the integrals: example

■ Function $f(x) = \cos(\pi x^2/2)$

- $N_{GP} = 1$: error $\approx 28.22\%$
- $N_{GP} = 2$: error $\approx 11.04\%$
- $N_{GP} = 3$: error $\approx 1.14\%$
- $N_{GP} = 4$: error $\approx 0.14\%$
- $N_{GP} = 5$: error $\approx 0.01\%$

■ Function $f(x) = x \sin(\pi x)$

- $N_{GP} = 1$: error $\approx 100.00\%$
- $N_{GP} = 2$: error $\approx 76.05\%$
- $N_{GP} = 3$: error $\approx 12.07\%$
- $N_{GP} = 4$: error $\approx 0.80\%$
- $N_{GP} = 5$: error $\approx 0.03\%$



Evaluation of the integrals II

- Consider:
$$\int_V [\mathbf{B}]^T [\mathbf{D}] [\mathbf{B}] dV = \sum_{e=1}^{N_e} \int_{V_e} [\mathbf{B}]^T [\mathbf{D}] [\mathbf{B}] dV$$

- Transpose to the parametric space (2D example)

$$\int_{V_e} [\mathbf{B}(\xi, \eta)]^T [\mathbf{D}] [\mathbf{B}(\xi, \eta)] dV = \int_{-1}^1 \int_{-1}^1 [\mathbf{B}(\xi, \eta)]^T [\mathbf{D}] [\mathbf{B}(\xi, \eta)] \det([\mathbf{J}]) d\xi d\eta$$

- Finally:

$$[\mathbf{K}] = \int_V [\mathbf{B}]^T [\mathbf{D}] [\mathbf{B}] dV = \sum_{e=1}^{N_e} \sum_{GP=1}^{N_{GP}} [\mathbf{B}^e(\xi_{GP}, \eta_{GP})]^T [\mathbf{D}] [\mathbf{B}^e(\xi_{GP}, \eta_{GP})] \det([\mathbf{J}^e(\xi_{GP}, \eta_{GP})]) w_{GP}$$

Evaluation of the integrals III

- If $N(\xi, \eta) = P_p$ is a polynomial of order p , then $[J] = P_{\dim(p-1)}$,
 $[B] = \frac{P_{2(p-1)}}{Q_{\dim(p-1)}}$
- **Remark I:** Gauss quadrature is exact for $p = 1$ and approximate if $p > 1$.
- **Remark II:** Stress and strains are exactly evaluated only in Gauss points, in all other points they are extrapolated/interpolated
- **Remark III:** 1 GP for linear triangle, 3 GP for quadratic triangle, 4 GP for bilinear quadrilateral element, 9 GP for quadratic quadrilateral, etc.
- **Remark IV:** Underintegration may lead to zero-energy deformation modes (which are often stabilized in FE software)

Evaluation of the integrals: quadrilateral 2D element

■ Shape functions:

$$N_1 = \frac{1}{4}(1 - \xi)(1 - \eta), \quad N_2 = \frac{1}{4}(1 + \xi)(1 - \eta)$$

$$N_3 = \frac{1}{4}(1 + \xi)(1 + \eta), \quad N_4 = \frac{1}{4}(1 - \xi)(1 + \eta)$$

■ Shape function derivatives:

$$N_{1,\xi} = -\frac{1}{4}(1 - \eta), \quad N_{2,\xi} = \frac{1}{4}(1 - \eta)$$

$$N_{3,\xi} = \frac{1}{4}(1 + \eta), \quad N_{4,\xi} = -\frac{1}{4}(1 + \eta)$$

$$N_{1,\eta} = -\frac{1}{4}(1 - \xi), \quad N_{2,\eta} = -\frac{1}{4}(1 + \xi)$$

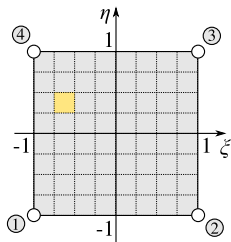
$$N_{3,\eta} = \frac{1}{4}(1 + \xi), \quad N_{4,\eta} = \frac{1}{4}(1 - \xi)$$

■ Determinant of Jacobian ($dA = \det [J] d\xi d\eta$):

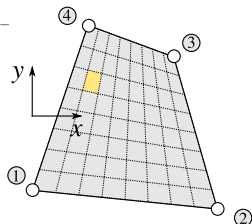
$$\det([J]) =$$

$$\frac{1}{16} [(1 - \eta)(x_2 - x_1) + (1 + \eta)(x_3 - x_4)]((1 + \xi)(y_3 - y_2) + (1 - \xi)(y_4 - y_1)) -$$
$$- ((1 - \eta)(y_2 - y_1) + (1 + \eta)(y_3 - y_4))((1 + \xi)(x_3 - x_2) + (1 - \xi)(x_4 - x_1))]$$

Parametric space



Physical space



Evaluation of the integrals: quadrilateral 2D element

■ Shape functions:

$$N_1 = \frac{1}{4}(1 - \xi)(1 - \eta), \quad N_2 = \frac{1}{4}(1 + \xi)(1 - \eta)$$

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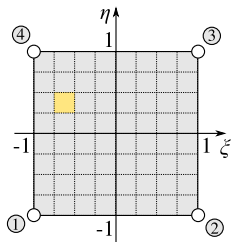
■ Determinant of Jacobian ($dA = \det[\mathbf{J}]d\xi d\eta$):

$$\det([\mathbf{J}]) =$$

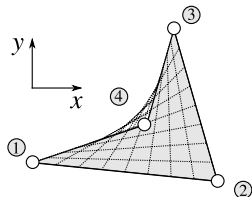
$$\frac{1}{16} [((1 - \eta)(x_2 - x_1) + (1 + \eta)(x_3 - x_4))((1 + \xi)(y_3 - y_2) + (1 - \xi)(y_4 - y_1)) - ((1 - \eta)(y_2 - y_1) + (1 + \eta)(y_3 - y_4))((1 + \xi)(x_3 - x_2) + (1 - \xi)(x_4 - x_1))]$$

■ **Warning:** to ensure $\det([\mathbf{J}]) > 0$ the element should remain convex

Parametric space



Physical space



Problem: Find $[\mathbf{u}]$ such that $[\mathbf{K}][\mathbf{u}] = [\mathbf{f}]$, i.e. $[\mathbf{u}] = [\mathbf{K}]^{-1} [\mathbf{f}]$

■ Iterative solvers

The solution is approached iteratively, does not require much memory, restrictions to matrix type, sensitive to matrix conditioning, a preconditioner is often needed.

- Gauss-Seidel method (GS)
- Conjugate gradient method (CG)
- Generalized minimum residual method (GMRES)
- ...

■ Direct solvers

The solution is provided directly, no restrictions on matrix type, less sensitive to matrix conditioning, based on LU or Cholesky decomposition

- Frontal
- Sparse direct
- ...

Example

- 3 bars in 2D
- 3 elements, 3 nodes, 6 dofs

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Thank you for your attention!
